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CALCULATION SHEET

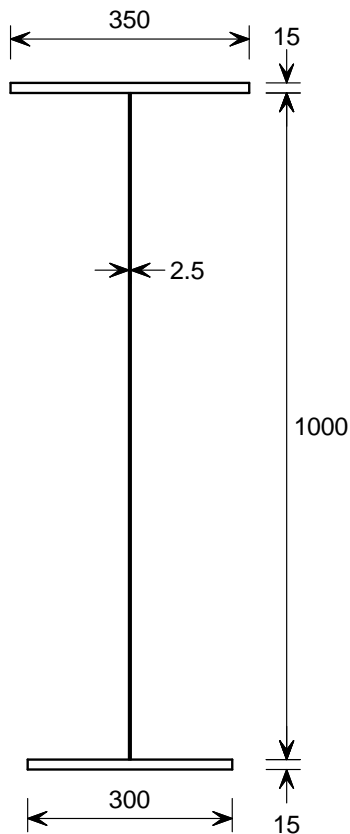
Job No.	Sheet 1 of 7	Rev A
Title		
Subject SINBEAM		
Client	Made by DGB	Date Feb 2018
	Checked by	Date

Design in Accordance with EN 1993-1-5 (Annex D) and EN 1993-1-3

Design combinations of actions from EN 1990

UK National Annex provisions adopted in all cases

Beam cross-section



Top flange area = $15 \times 350 = 5250 \text{ mm}^2$
 Bottom flange area = $15 \times 300 = 4500 \text{ mm}^2$
 Web area = $1000 \times 2.5 \times \frac{178}{155} = 2871 \text{ mm}^2$
 Note the 178/155 allows for the corrugations.
 Total area = 12621 mm^2

Self weight = $12621 \times 7850 \times 9.81/10^9 = 0.97 \text{ kN/m}$

Beam loading

Full UDL $g_k = 10 \text{ kN/m}$

$q_k = 10 \text{ kN/m}$

Central point load $G_k = 50 \text{ kN}$

$Q_k = 75 \text{ kN}$

Total permanent action = $10 + 0.97 = 10.97 \text{ kN/m}$

Design combination of actions using expression 6.10b

$$\text{ULS UDL} = 1.35 \times 0.925 \times 10.97 + 1.5 \times 10 = 28.7 \text{ kN/m}$$

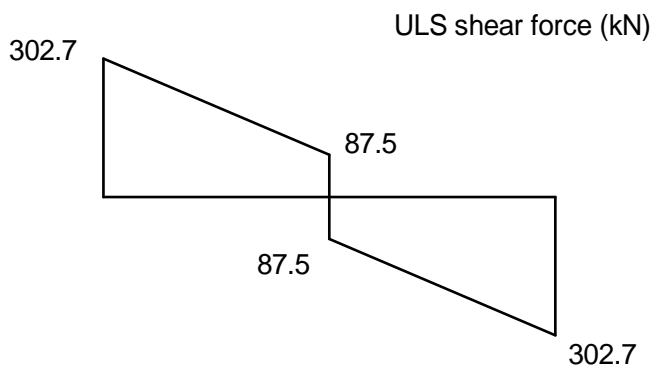
$$\text{ULS point load} = 1.35 \times 0.925 \times 50 + 1.5 \times 75 = 174.94 \text{ kN}$$

15 m span

Left reaction

$$= 28.7 \times \frac{15}{2} + \frac{174.94}{2} = 302.72 \text{ kN}$$

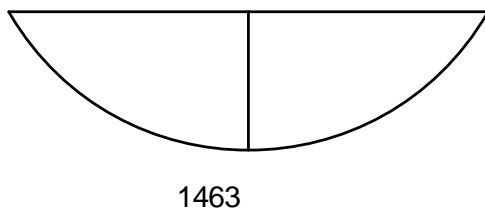
SF Diagram



Central Bending Moment

$$= 28.7 \times 15^2/8 + 174.94 \times \frac{15}{4} = 1463 \text{ kNm}$$

ULS bending moment (kNm)

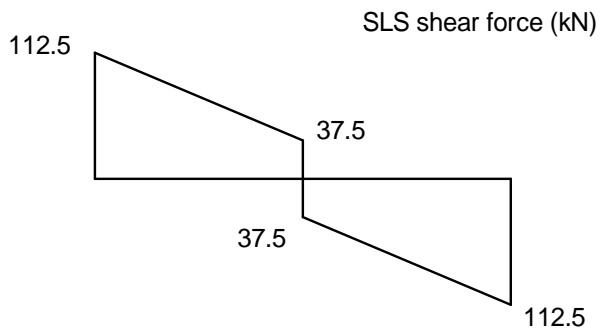


SLS forces and moments

$$\text{SLS UDL} = 10 \text{ kN/m}$$

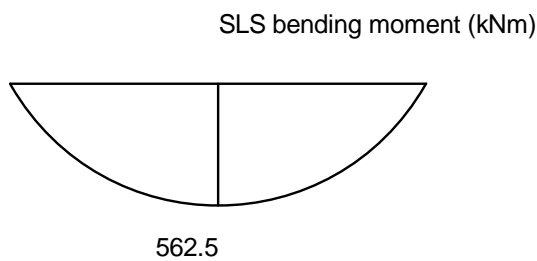
$$\text{SLS point load} = 75 \text{ kN}$$

$$\text{SLS end reaction} = 10 \times \frac{15}{2} + \frac{75}{2} = 112.5 \text{ kN}$$



SLS Bending Moment

$$10 \times \frac{15^2}{8} + 75 \times \frac{15}{4} = 562.5 \text{ kNm}$$



Cross-section Inertia

$$\text{Web area} = 2.5 \times 1000 = 2500 \text{ mm}^2$$

$$\text{Total area} = 2500 + 5250 + 4500 = 12250 \text{ mm}^2$$

For neutral axis:

$$5250 \times 1022.5 + 2500 \times 515 + 4500 \times 7.5 = \bar{y} \times 12250$$

$$\bar{y} = 546 \text{ mm from bottom}$$

$$\begin{aligned} \text{Inertia} &= \frac{350 \times 15^3}{12} + 350 \times 15 \times (1022.5 - 546)^2 \\ &+ \frac{2.5 \times 1000^3}{12} + 2.5 \times 1000 \times (515 - 546)^2 \\ &+ \frac{300 \times 15^3}{12} + 300 \times 15 \times (7.5 - 546)^2 \\ &= 2.707 \times 10^9 \text{ mm}^4 \end{aligned}$$

Central Deflections

Bending deflections

$$\text{Point load} = \frac{75 \times 10^3 \times 15000^3}{48 \times 210 \times 10^3 \times 2.707 \times 10^9} = 9.28 \text{ mm}$$

$$\text{UDL} = \frac{5 \times 10 \times 15000^4}{384 \times 210 \times 10^3 \times 2.707 \times 10^9} = 11.60 \text{ mm}$$

Shear deflections

$$\text{Point load} = \frac{75 \times 10^3 \times 15000}{4 \times 81000 \times 1000 \times 2.5 \times (155/178)} = 1.60 \text{ mm}$$

$$\text{UDL} = \frac{10 \times 10^3 \times 15 \times 15000}{8 \times 81000 \times 1000 \times 2.5 \times (155/178)} = 1.60 \text{ mm}$$

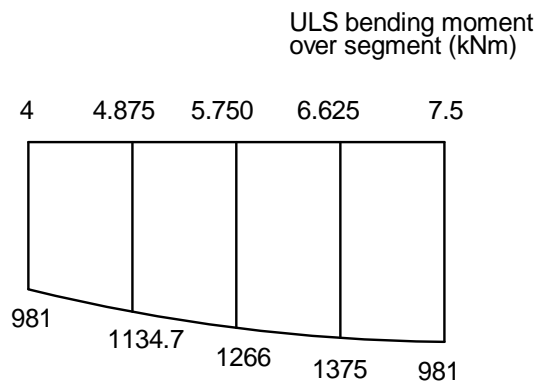
$$\text{Total} = 9.28 + 11.60 + 1.60 + 1.60 = 24.1 \text{ mm}$$

Top flange verification

Between restraints at 4 m and 7.5 m

Length = 3500 mm

Bending moment diagram over segment:



$$\text{At 4 m BM} = 302.72 \times 4 - 28.7 \times \frac{4^2}{2} = 981 \text{ kNm}$$

$$\text{At 4.875 m BM} = 302.72 \times 4.875 - 28.7 \times \frac{4.875^2}{2} = 1134.7 \text{ kNm}$$

$$\text{At 5.75 m BM} = 302.72 \times 5.75 - 28.7 \times \frac{5.75^2}{2} = 1266 \text{ kNm}$$

$$\text{At 6.625 m BM} = 302.72 \times 6.625 - 28.7 \times \frac{6.625^2}{2} = 1375 \text{ kNm}$$

$$C_1 = \sqrt{\frac{35 \times 1463^2}{1463^2 + 9 \times 1134.7^2 + 16 \times 1266^2 + 9 \times 1375^2}} = 1.153$$

$$k_c = \frac{1}{\sqrt{C_1}} = \frac{1}{\sqrt{1.153}} = 0.93$$

Minor axis inertia of flange

$$= \frac{15 \times 350^3}{12} = 53.6 \times 10^6 \text{ mm}^4$$

$$\text{Radius of gyration} = \sqrt{\frac{53.6 \times 10^6}{5250}} = 101 \text{ mm}$$

Steel strength – according to UK NA is from product standard

$$15 \text{ mm} < 16. \text{ so } f_y = 355 \text{ N/mm}^2$$

BS EN 10025-2

$$\varepsilon = \sqrt{\frac{235}{355}} = 0.814$$

$$\lambda_1 = 93.9 \varepsilon = 93.9 \times 0.814 = 76.4$$

$$\bar{\lambda}_f = \frac{0.93 \times 3500}{10.1 \times 76.4} = 0.42$$

For minor axis buckling, $f_y < 460 \text{ N/mm}^2$ and $t < 40 \text{ mm}$

Use curve 'c'

$$\alpha = 0.49$$

$$\text{Then } \phi = 0.5 [1 + 0.49 (0.42 - 0.2) + 0.42^2] = 0.64$$

$$\text{Then } \chi = \frac{1}{0.64 + \sqrt{0.64^2 - 0.42^2}} = 0.89$$

Shear lag

$$b_o = 350/2 = 175 \text{ mm}$$

$$\ell_o/50 = 15000/50 = 300$$

$175 < 300$, so shear lag may be neglected $\beta = 1.0$

Buckling factor k_σ

Note Zeman do not use 1993-1-5 Annex D, Clause D.2.1(2)

Zeman use Table 4.2 of 1993-1-5 for k_σ

Because $\beta = 1.0$

$$\frac{\sigma_2}{\sigma_1} = 1.25 (1 - 0.2) = 1.25 (1 - 0.2) = 1.0$$

From Table 4.2, if $\psi = 1.0$, $k_\sigma = 0.43$

then:

$$\begin{aligned} \bar{\lambda}_p &= \frac{\bar{b}/t}{28.4 \varepsilon \sqrt{k_\sigma}} & \bar{b} &= \frac{b - 0.5 \times 9}{2} \\ & & &= \frac{350 - 0.5 \times 40}{2} \\ & & &= 165 \text{ mm} \end{aligned}$$

$$\text{so } \bar{\lambda}_p = \frac{165/15}{28.4 \times 0.814 \times \sqrt{0.43}} = 0.725$$

because $0.725 < 0.748$, $\rho = 1.0$

$$A_{\text{eff}} = 1.0 \times A_c = 5250 \text{ mm}^2$$

Flange resistance

$$= 0.89 \times 5250 \times 355 \times 10^{-3} = 1659 \text{ kN}$$

Force in top flange

Maximum moment in segment = 1463 kNm

6.3.2.4

Table 6.2

Table 6.1

1993-1-5 3.1

$$\text{Force} = \frac{1463 \times 10^3}{1000 + 7.5 + 7.5} = 1441 \text{ kN}$$

ULS axial force

$$= 135 \times 0.925 \times 100 + 1.5 \times 50 = 199.9 \text{ kN}$$

Proportion in top flange

$$= 199.9 \times 5250 / (5250 + 4500) = 107.6$$

$$\text{Total axial force} = 1441 + 107.6 = 1549 \text{ kN}$$

$$\text{Utilisation} = 1549 / 1659 = 0.93, < 1.0, \text{ ok}$$

Bottom flange verification

$$b_o = 300 / 2 = 150$$

$$\ell_e / 50 = 15000 / 50 = 300$$

$300 > 150$, so shear lag may be neglected

Maximum moment = 1463 kNm

$$\text{Force in flange} = \frac{-1463 \times 10^3}{1000 + 7.5 + 7.5} = -1441 \text{ kN (tension)}$$

$$\text{Compressive force} = 199.9 \times 4500 / (5250 + 4500) = 92.3 \text{ kN}$$

$$\text{Nett tension} = -1441 + 92.3 = -1349 \text{ kN}$$

$$\text{Resistance} = 4500 \times 355 \times 10^{-3} = 1598 \text{ kN}$$

$$\text{Utilisation} = 1349 / 1598 = 0.84$$

Web verification

Maximum shear = 302.7 kN

Global buckling

Separately, the following numerical integrations have determined the following values:

Web	D_x	D_z
1.5	56210.9	6.31×10^7
2	133240	8.41×10^7
2.5	260236	1.05×10^8
3	449687	1.26×10^8

For 2.5 mm web

$$\tau_{cr,g} = \frac{32.4}{2.5 \times 1000^2} \sqrt[4]{260236 \times 1.05 \times 10^8}$$

$$= 303.6 \text{ N/mm}^2$$

In accordance with Note 1 to Table 3.1a of EN 1993-1-3, the yield strength is factored by 0.9.

The design yield strength is therefore $0.9 \times 355 = 319.5 \text{ N/mm}^2$

$$\bar{\lambda}_{c,g} = \sqrt{\frac{319.5}{303.6 \sqrt{3}}} = 0.78$$

1993-1-5
Annex D.2.2

$$\chi = \frac{1.5}{0.5 + 0.78^2} = 1.35 \text{ but } < 1$$

$$\text{So } \chi = 1.0$$

Local buckling

$$w = 155 \text{ mm}$$

$$s = 178 \text{ mm}$$

$$\tau_{cr,1} = \left(5.34 + \frac{40 \times 178}{1000 \times 2.5} \right) \left(\frac{\pi^2 \times 210,000}{12(1 - 0.3^2)} \right) \left[\frac{2.5}{178} \right]^2$$

$$= 8.188 \times 189800 \times 1.9726 \times 10^{-4}$$

$$= 306.6 \text{ N/mm}^2$$

$$\bar{\lambda}_{cr,1} = \sqrt{\frac{319.5}{306.6 \times \sqrt{3}}} = 0.78$$

$$\chi_{u,1} = \frac{1.15}{0.9 + 0.78} = 0.68$$

$$V_{bw,Rd} = 0.68 \times \frac{319.5}{\sqrt{3}} \times 1000 \times 2.5 \times 10^{-3} = 316 \text{ kN}$$

$$\text{Utilisation} = 302.7/316 = 0.96 < 1.0, \text{ OK}$$